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Focal surface attitude detection for LAMOST

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Received 2020 February 24; accepted 2020 May 22

Abstract With telescope apertures becoming larger and larger, the deployment of large-field telescopes is becoming increasingly popular. However, optical path calibration is necessary to ensure the image quality of large-field and large-diameter telescopes. In particular, focal plane attitude calibration is an essential optical path calibration technique that has a direct impact on image quality. In this paper, a focal plane attitude detection method using eight acquisition cameras is proposed based on the calibration requirements of the wide-field telescope, LAMOST (Large Sky Area Multi-Object Fiber Spectroscopic Telescope). Comparison of simulation and experimental results shows that the detection accuracy of the proposed method can reach 30 arcsec. With additional testing and verification, this method could be used to facilitate regular focal plane attitude calibration for LAMOST as well as other large-field telescopes.

Key words: methods: data analysis - techniques: high angular resolution - techniques: image processing

1 INTRODUCTION

A telescope is an essential tool for exploring the universe. To achieve highly accurate celestial body information, image quality is especially crucial. The observatory site, aperture, and optical path calibration have a significant impact on the image quality of the telescope. Focal plane attitude calibration, as an element of optical path calibration, is particularly important (McLeod 1996; Schechter & Levinson 2011). The condition of the focal plane attitude directly affects the shape of the image.

The Large Sky Area Multi-Object Fiber Spectroscopic Telescope (LAMOST) is the largest-diameter and highest-spectral-acquisition-rate optical telescope with wide field of view (FOV). The telescope consists of a reflective, active, aspheric Schmidt corrector plate (Ma); a spherical primary mirror (Mb); and a focal surface. The LAMOST telescope has a 5° FOV, and the focal plane diameter is 1.75 m (Cui et al. 2012; Zhao et al. 2012). Deviation of the attitude from the ideal focal plane will cause image spot distortion, resulting in energy loss into the optic fibre.

The image quality required is that 80% of the optical energy should be within 1.5 arcsec (Su & Cui 2004).

According to simulation data, it will cause the image size of the LAMOST telescope to increase by 0.1 arcsec if focal plane rotates around the horizon axis about 54 arcsec in the reverse direction or 2.4 arcmin in the forward direction or 2.3 arcmin around the vertical axis. To avoid unnecessary energy loss, focal plane attitude calibration is required regularly.

Large FOV telescopes currently built include the Sloan Digital Sky Survey (SDSS) (York et al. 2000), VLT Survey Telescope (VST) (Shanks et al. 2015), Two Micron All-Sky Survey (2MASS) (Skrutskie et al. 2006), Visible and Infrared Survey Telescope for Astronomy (VISTA) (Sutherland et al. 2015), and the Wide-field Infrared Survey Explorer (WISE) (Wright et al. 2010). Upcoming construction includes the Large Synoptic Survey Telescope (LSST) (Tyson 2002) and the Wide Field Infrared Survey Telescope (WFIRST) (Content et al. 2013). Focal plane attitude detection is also essential for these telescopes.

Traditional focal plane attitude detection uses the self-collimation method (Hemayed 2003; Li et al. 2016), which requires active optical self-collimation technology. This approach is affected by atmospheric visibility and suffers from low detection accuracy. Some telescopes

adopt alternative approaches, such as the focal surface camera used by the VISTA telescope of the European Southern Observatory. VISTA's focal surface camera captures bright star images, using image size to detect the focal plane attitude. This technique requires 180° focal plane rotation and a large-area charged couple device (CCD) (Sutherland et al. 2015; Terrett & Sutherland 2010; Dalton et al. 2006). However, LAMOST is a spectral survey telescope with 4000 optical fibres. This method cannot be applied in LAMOST.

Aiming at the characteristics of the wide FOV of LAMOST, it is proposed to use the existing eight acquisition cameras on the focal plane to obtain a series of defocused spot images on either side of the ideal focus surface and to then calculate the focal plane attitude based on the defocus values from all of the cameras.

Section 2 of this paper introduces the principles of our detection method. The simulation experiment and error analysis presented in Section 3 verify the feasibility of our approach, while Section 4 introduces the experimental results that we obtained using our method of detecting the focal plane attitude of the LAMOST telescope. In the final section, we summarise our work and discuss the application of our method in the future.

2 DETECTION METHOD

The LAMOST optical system includes mirror Ma, mirror Mb, and focal surface, as shown in Figure 1. Ma $(5.72 \text{ m} \times 4.40 \text{ m})$ consists of 24 hexagonal segments with a diagonal of 1.1 m. Mb $(6.67 \text{ m} \times 6.05 \text{ m})$ has 37 hexagonal segments with a diagonal diameter of 1.1 m. The Mb has a radius of curvature of 40 m and a system focal length of 20 m (Su et al. 2012). The focal surface has 1.75 meters diameter and eight acquisition cameras, the focal surface picture and arrangement of eight acquisition cameras are shown in Figure 2.

According to geometrical optics, the image has a pupil shape in the case of long-distance defocusing, and the image size increases linearly with the defocus distance (Born & Wolf 2003).

Taking the positive direction of defocus (toward Ma) as an example, the position of the focus can be calculated by obtaining a series of defocus positions x_n and the size y_n of the defocused image

$$[(x_1, y_1) (x_2, y_2) (x_3, y_3) \cdots (x_n, y_n)].$$

Because the image size increases linearly with the defocus distance, the relationship of image size and defocus distance satisfies Equation (1).

$$\begin{bmatrix} y_1 \\ y_2 \\ y_3 \\ \vdots \\ y_n \end{bmatrix} = a_1 \times \begin{bmatrix} x_1 - x_{01} \\ x_2 - x_{01} \\ x_3 - x_{01} \\ \vdots \\ x_n - x_{01} \end{bmatrix},$$
(1)

where x_{01} is the theoretical focus position estimate value.

We can obtain the value of a_1 and x_{01} by linear fitting with the least squares method.

Furthermore, because of aberration, we need to consider the case of negative direction defocus (toward Mb) to get the estimated value x_{02} of the theoretical focus position. The final focus position is given by the average of two independent x values, $x_{03} = (x_{01} + x_{02})/2$, and the deviation of the focus position $\Delta f = x_{03} - x_{\text{focal}}$ is calculated where x_{focal} is the ideal focus position.

To obtain the tilt angle $\Delta \theta$ of the focal surface relative to the two detector directions, we need to compare the relative deviation $\Delta f' = \Delta f_1 - \Delta f_2$ of the two detectors and relative distance D between the two detectors.

$$\Delta \theta = \Delta f' / D \,. \tag{2}$$

Further, before and after the focal surface, we explore the light intensity distribution of the image, as shown in Figure 3, where P_1 and P_2 are two surfaces with defocus distance L.

The distribution of light complex amplitude before passing reflector M_b is

$$\psi(r) = A_0 \exp(i\frac{2\pi}{\lambda}w(r)).$$
(3)

According to the Fraunhofer diffraction approximation, the light complex amplitudes of the images on P_1 and P_2 , respectively, are Goodman (2005):

$$A_{P1}(r) = \psi(r) \otimes \left\{ \frac{1}{i\lambda(f-l)} \exp[i\pi \frac{r^2}{\lambda(f-l)}] \right\}, \quad (4)$$

$$A_{P2}(r) = \psi(r) \otimes \left\{ \frac{1}{i\lambda(f+l)} \exp[i\pi \frac{r^2}{\lambda(f+l)}] \right\}.$$
 (5)

Obtained by geometric optical approximation, the relationship of the light intensities in the P_1 and P_2 planes is Roddier (1988):

$$\frac{CI_{P2}(r) - DI_{P1}(-r)}{CI_{P2}(r) + DI_{P1}(-r)} = S,$$
(6)

$$S = \frac{C-D}{C+D} + \frac{2D}{C+D} \frac{f(f-l)}{l} \times [\nabla^2 w(\frac{fr}{l}) - \frac{\partial}{\partial n} w(\frac{fr}{l}) \delta_c],$$
(7)



Fig. 1 LAMOST: a general view. LAMOST is a special reflecting Schmidt telescope with a 4 m aperture and a 5° FOV. It has a focal length of 20 m and an f-ratio of 5. Its optical axis is fixed in the meridian plane and is tilted by 25° to the horizontal. Celestial objects are observed for 1.5 hours as they cross the meridian. During the observation period only the alt-azimuth mount of the reflecting Schmidt correcting mirror (Ma) and the focal surface do the tracking. The survey area to be observed is $-10^{\circ} \le \delta \le +90^{\circ}$.



Fig. 2 The picture shows the focal surface of LAMOST; The left picture is the photo of the focal surface which is mounted with eight acquisition cameras, the right image shows the relative positions of eight cameras, four inner cameras in the 3-degree FOV in *blue* and four outer cameras in the 5-degree FOV in *green*.



Fig. 3 Light reflected by the Mb mirror image onto the surface F. P_1 and P_2 are surfaces with the same defocus distance. The optic construction is similar to the curvature wavefront sensor, but our system moves the focal surface and variable defocus distance.

where $I_{P1}(-r)$ and $I_{P2}(r)$ are the symmetric points light intensities of the P_1 and P_2 planes, respectively, and C, Dare constants, which satisfy

$$\frac{D}{C} = (f+l)(f-l).$$
(8)

It can be seen that the intensity distribution at the same distance before and after the focus is asymmetrical, unless wavefront curvature $\nabla^2 w$ and wavefront radial tilt $\frac{\partial w}{\partial n}$ are zero.

Not only does the light intensity distribution differ at a given distance before and after the focal surface, but also the particular pupil shape creates difficulty in measuring the image sizes. A high-precision detection method is needed to extract the size of the defocus image.

In a comparison of the ellipse fitting algorithm (Fitzgibbon et al. 1999; Gander et al. 1994), multi-peak Gaussian fitting algorithm (Guo 2011; Wan et al. 2018), average spacing algorithm (ASA), and peak spacing algorithm (Hu et al. 2020), the ASA has the highest detection accuracy and is least influenced by stellar brightness change. This algorithm can guarantee the accuracy of our image size detection. Details of the average spacing algorithm are given in Section 3.1.

3 SIMULATION AND ANALYSIS

Before the experiment, we used Zemax software for simulation analysis to verify the feasibility of our method (Geary 2002). We built the LAMOST simulation software optical path to generate the defocusing images and calculated the defocus image sizes using the ASA.

Using the optical path of the LAMOST telescope, ray tracing can be used to obtain the theoretical spot size. When the star is near the culmination of the 30° survey area, we detect the defocus image size from the negative defocusing 25 mm position to the positive defocusing 25 mm position and select the 100% light energy-containing region as the spot size, as shown in Figure 4.

We can see that the size of the image near the focus changes nonlinearly. To ensure linear behaviour, we chose defocus distances of 20–25 mm in the simulation analysis.

3.1 Average Spacing Algorithm

The ASA is used to estimate the image size, which is then simulated by the Zemax software. First, we choose a clear image and abstract the image area $k \times k$, which is an intensity distribution matrix I. I(x, y) is light intensity in (x, y). Second, we find the maximum light intensity matrix A and minimum light intensity matrix B of light intensity distribution matrix I. Values A(x) and B(x) are the maximum and minimum values of row x of I, respectively:

$$A = \begin{bmatrix} A(1) \\ A(2) \\ A(3) \\ \vdots \\ A(k) \end{bmatrix} = \max \begin{bmatrix} I(1,1) & I(1,2) & \cdots & I(1,k) \\ I(2,1) & I(2,2) & \cdots & I(2,k) \\ I(3,1) & I(3,2) & \cdots & I(3,k) \\ \vdots & \vdots & \ddots & \vdots \\ I(k,1) & I(k,2) & \cdots & I(k,k) \end{bmatrix},$$
(9)

$$B = \begin{bmatrix} B(1) \\ B(2) \\ B(3) \\ \vdots \\ B(k) \end{bmatrix} = \min \begin{bmatrix} I(1,1) \ I(1,2) \ \cdots \ I(1,k) \\ I(2,1) \ I(2,2) \ \cdots \ I(2,k) \\ I(3,1) \ I(3,2) \ \cdots \ I(3,k) \\ \vdots \ \vdots \ \ddots \ \vdots \\ I(k,1) \ I(k,2) \ \cdots \ I(k,k) \end{bmatrix}.$$
(10)

Using average value B_{ave} of B as background noise, $C = A - B_{\text{ave}}$ represents new light intensity distribution. I_{ave} is the average value of C.

Using Lagrangian interpolation to improve detection accuracy, we find the values x_0 and x_1 that satisfy $A(x_0) = a$ and $A(x_1) = a$, where $a = 0.5 \times I_{\text{ave}} + B_{\text{ave}}$. $L = |x_0 - x_1|$ is the size of image Figure 5.

3.2 Focal Point and Image Size Detection for Constructing the Look-up Table

Using a simulated optical path of a 30° survey area with the central star in the 5-degree hour angle position before the culminant, we simulated the images at defocus distances of 20, 21, 22, 23, 24 and 25 mm in all eight acquisition cameras. We used the Zemax software and the ASA to calculate the image size, then evaluated the focus position. A simulation image of central field of view by Zemax software is shown in Figure 6. The test result of the eight fields of view, corresponding to LAMOST's actual eight acquisition cameras, is shown in Tables 1, 2, 3 and 4.

The ideal focal position based on LAMOST's unique design is an average result derived from different survey areas (Cui et al. 2012). According to the results of the image size calculations using Zemax software and the ASA, the focal point estimation position using our method experienced have some deviation from the focal surface. Nevertheless, the estimate of focal position generated using Zemax software and the ASA is close enough to focal surface to construct a look-up table (Lucente 1993) for experimental testing.

Comparing the evaluation value of the focus point calculated by our algorithm and the Zemax simulation,



Fig.4 The image size changes with the defocus distance. Near the focal surface, spot size change is small. On the 1 cm to 15 cm defocus distance scale, image size decreases with the defocus distance but not linearly. When the defocusing distance exceeds 15 cm, spot size decreases linearly with the defocus distance.



Fig. 5 The image process by the average spacing algorithm. $Red' \triangle'$ is maximum light intensity A, orange ' \circ' is minimum light intensity B, and the blue line is $a = 0.5 \times I_{ave} + B_{ave}$. x_0 and x_1 are the points of intersection of the blue and red lines.

which is shown in Table 5, the detection accuracy of the four detectors in the inner circle is within 70 μ m while the outer ring is within 90 μ m. The detection accuracy of the tilt angle is 30".

When the central star is on the culminant, the defocused image size of different fields is shown in Table 6. In the experiment, when we get the star image size to defocus distance, we use the linear fitting which shows a good linear and Lagrangian interpolation method to process data and get the focus position of different

cameras. We use the look up table method to compare the focal point position of the different cameras with the simulation result to get the defocus value of the different cameras. Then according to Equation (2), we can get the focal surface tilt value.

3.3 Effect of Focal Plane Rotation on Detection Accuracy

The azimuthal and vertical angles of Ma and the rotation angle of the focal surface change with telescope tacking.

Table 1 Image size estimated by Zemax software. FourFOVs correspond to the four acquisition cameras in theinner circle of the focal plane (unit: mm).

Distance	-1.5/-1.5	1.5/1.5	1.5/-1.5	-1.5/1.5
-25	5.914	5.804	5.794	6.027
-24	5.671	5.577	5.558	5.793
-23	5.429	5.350	5.322	5.558
-22	5.187	5.123	5.086	5.324
-21	4.944	4.896	4.850	5.089
-20	4.702	4.669	4.615	4.855
20	5.018	4.441	4.858	4.568
21	5.261	4.668	5.094	4.802
22	5.503	4.895	5.330	5.037
23	5.745	5.122	5.566	5.271
24	5.987	5.349	5.802	5.506
25	6.230	5.576	6.038	5.741

Table 2 Image size estimated by Zemax software. FourFOVs correspond to the four acquisition cameras in theouter circle of the focal plane (unit: mm).

Distance	-2.5/0	2.5/0	0/-2.5	0/2.5
-25	6.352	5.972	5.143	5.172
-24	6.098	5.734	4.930	4.972
-23	5.844	5.495	4.718	4.772
-22	5.590	5.256	4.505	4.572
-21	5.336	5.018	4.293	4.372
-20	5.082	4.779	4.080	4.172
20	5.137	4.793	4.430	3.838
21	5.390	5.031	4.643	4.038
22	5.644	5.270	4.855	4.238
23	5.897	5.508	5.068	4.438
24	6.150	5.747	5.281	4.638
25	6.403	5.986	5.493	4.839

Table 3Image size detected by the average spacingalgorithm. Four FOVs correspond to the four acquisitioncameras in the inner circle of the focal plane (unit: pixel).

Distance	-1.5/-1.5	1.5/1.5	1.5/-1.5	-1.5/1.5
-25	205.243	200.383	191.581	202.136
-24	196.597	192.392	183.604	194.741
-23	188.668	185.286	175.271	187.004
-22	179.767	177.522	168.011	178.980
-21	170.806	169.555	160.568	170.602
-20	162.076	160.525	152.003	162.864
20	171.494	149.638	158.530	152.273
21	180.124	157.081	164.968	159.770
22	188.921	165.174	172.971	167.813
23	196.941	171.901	180.610	175.601
24	205.350	180.154	188.978	183.309
25	214.305	188.538	196.573	191.199

Since any of them may cause image size detection error, it is crucial to analyse their effects. Focal plane rotation angle θ changes with hour angle t_1 satisfies (Su & Wang 1997).

$$\sin \theta = \left[-\cos(A + \phi) + \sin(A + \phi) \right] \times \tan\left(\frac{1}{2}\hat{AS}\right) \cos ZAS \sin(t_1)$$
(11)

Table 4Image size detected by the average spacingalgorithm. Four FOVs correspond to the four acquisitioncameras in the outer circle of the focal plane (unit: pixel).

Distance	-2.5/0	2.5/0	0/-2.5	0/2.5
-25	289.992	243.324	241.658	239.024
-24	278.401	233.247	231.249	230.774
-23	267.659	222.918	221.831	220.711
-22	256.843	213.327	212.379	211.821
-21	245.448	202.002	201.395	201.511
-20	233.306	192.126	192.480	193.500
20	236.180	188.967	209.502	182.270
21	247.978	198.930	220.209	191.721
22	258.419	209.706	229.280	200.911
23	269.939	220.155	239.908	210.524
24	281.076	229.203	249.292	219.650
25	292.098	239.463	259.509	229.589

Table 5 Focal point evaluation and deviation value using Zemax software and the average spacing algorithm (ASA) for eight fields of view. A negative value indicates evaluation at the Mb side and a positive value indicates evaluation at the Ma side. The results reveal that the ASA and Zemax software produce similar results, and the ASA can be used to evaluate the ideal focal point position (unit: μ m).

Field	Zemax software	ASA	Deviation
-1.5/-1.5	-654.6	-684.6	-30.0
1.5/1.5	501.0	571.7	70.7
1.5/-1.5	-516.0	-519.1	-3.1
-1.5/1.5	619.4	549.6	-70.2
-2.5/0	-139.3	-185.6	-46.3
2.5/0	-29.6	2.2	31.8
0/-2.5	-819.1	-799.5	20.4
0/2.5	842.3	755.4	-89.6

where A is 25° , ϕ is the astronomical site latitude, and $\frac{1}{2}\hat{AS}$ is the incident angle of the celestial body, as shown in Figure 7. Taking the 20° , 30° , and 40° survey areas as an example, the focal plane rotation angle is shown in the following Table 7.

The CCD acquisition dimensions of LAMOST are $2k \times 2k$, and the pixel size is 25 µm. Based on the focal plane rotation angle, every detector deviation can be calculated. In the 3.75° hour angle, the detector deviation is no more than 7 mm, which is far less than the detector size. Therefore, focal plane rotation has little effect on the image size.

3.4 The Influence of Ma Angle Change on Spot Size Detection

Because of the rotation of the earth, the change of the celestial body angle is $15 \operatorname{arcsec} s^{-1}$ as seen by the detector. Mirror Ma needs to control the vertical and azimuthal angles to keep tracking constant; however,



Fig.6 At left, a simulation image by Zemax software. At right, the middle section light intensity distribution of the simulation image. The light intensity distribution is uneven because every segmented mirror produces a pupil image in the detector. Moreover, because of the vignette effect, some regions have low light intensity.

Table 6 The defocus image size and focus position when the central star in the culminant position. Position (0) is focal estimate position with the central star in the culminant position, position (5) is focal estimate position with a central star at 5-degree hour angle position before culminant. It can been seen that the left position and right position have good symmetry and have a similar focal position to 5-degree hour angle (unit: mm).

DISTANCE/mm	1.5/1.5	1.5/-1.5	-1.5/1.5	-1.5/-1.5	2.5/0	-2.5/0	0/2.5	0/-2.5
20	4.5034	4.8744	4.5034	4.8744	4.8391	4.8391	3.9066	4.3684
21	4.7338	5.1113	4.7338	5.1113	5.0789	5.0789	4.1105	4.5782
22	4.9642	5.3482	4.9642	5.3482	5.3187	5.3187	4.3146	4.7882
23	5.1946	5.5851	5.1946	5.5851	5.5586	5.5586	4.5187	4.9981
24	5.4252	5.8220	5.4252	5.8220	5.7985	5.7985	4.7227	5.2080
25	5.6556	6.0590	5.6556	6.0590	6.0383	6.0383	4.9268	5.4180
-20	4.7472	4.6193	4.7472	4.6193	4.8143	4.8143	4.2552	4.0281
-21	4.9777	4.8561	4.9777	4.8561	5.0530	5.0530	4.4592	4.2380
-22	5.2081	5.0929	5.2081	5.0929	5.2919	5.2919	4.6633	4.4480
-23	5.4386	5.3297	5.4386	5.3297	5.5307	5.5307	4.8673	4.6579
-24	5.6691	5.5666	5.6691	5.5666	5.7696	5.7696	5.0714	4.8678
-25	5.8995	5.8035	5.8995	5.8035	6.0085	6.0085	5.2753	5.0777
Position(0)/µm	528.3	-535.3	528.3	-535.3	-9.7	-9.7	855.3	-810.4
Position(5)/µm	501.0	-516.0	619.4	-654.6	-29.6	-139.3	842.3	-819.1

Table 7 Focal plane attitude rotation angle of 20°, 30°, and 40° survey areas. The rotation angle increases with the hour angle, where the culmination position hour angle is 0. When the hour angle is small (≤ 3.75), the rotation angle is small. *t* is the hour angle and δ is the survey area.

t (°)	$\begin{array}{c} \delta = 20 \\ (^{\circ}) \end{array}$	$\begin{array}{c} \delta = 30 \\ (^{\circ}) \end{array}$	$\begin{array}{c} \delta = 40 \\ (^{\circ}) \end{array}$
$\pm 22.5 \\ \pm 11.25 \\ \pm 7.5 \\ \pm 3.75$	∓ 0.9887 ∓ 0.4895 ∓ 0.3258 ∓ 0.1627	∓ 1.2083 ∓ 0.5983 ∓ 0.3981 ∓ 0.1989	∓ 3.6119 ∓ 1.7889 ∓ 1.1905 ∓ 0.5946
0	0	0	0

this leads to a varying pupil shape and consequently a changing defocus image spot size. Vertical angle is $\angle AZN$; azimuthal angle is \widehat{ZN} . Different hour angle and survey area results are shown in Table 8. a celestial body observed near the culmination should be selected.

4 EXPERIMENT AND DISCUSSION

The result reveals that the experimental observation time needs to be controlled to within 20 minutes and that Before detection of the focal plane attitude of LAMOST, it is necessary to calibrate other parts of the optical path.



Fig. 7 Essential points and lines in the celestial sphere. O_0 is the spherical centre of the spherical mirror Mb. The optical axis O_0A passes through O_0 , lies in the meridian plane and is inclined at 25° to the horizontal. In a Schmidt system, the centre (vertex) of the reflecting correcting surface of Ma should be at O_0 , and the surface centre of Mb should be on the optical axis. After reflection by Ma, the ray SO_0 , which arrives at O_0 from the celestial object at the centre of the FOV, passes along the optical axis O_0A . The incident plane is the plane that includes the optical axis and the ray SO_0 . O_0N is the bisector of the angle between the optical axis O_0A and the beam SO_0 .



Fig. 8 Images from three cameras are presented. Left: No.0 camera. Centre: No.4 camera. Right: No.6 camera. The central dark spots are caused by shading of the focal plane and the central segmented mirror of Ma, which is masked.

Table 8 Deviation of detected spot size by hour angle in the survey areas of 20° , 30° , and 40° . The results show the deviation increases with the hour angle (unit: μ m).

	$\delta = 2$	0°	$\delta = 3$	80°	$\delta = 4$	40°
t	Horizontal	Vertical	Horizontal	Vertical	Horizontal	Vertical
$\pm 22.5^{\circ}$	-193.9	44.0	-218.8	50.9	-241.6	56.7
$\pm 11.25^{\circ}$	-44.9	10.9	-56.5	13.1	-62.6	14.6
$\pm 7.5^{\circ}$	-22.3	4.9	-25.3	5.8	-27.9	6.5
$\pm 3.75^{\circ}$	-5.6	1.2	-6.3	1.5	-7.0	1.6
0	0	0	0	0	0	0

The most important is coaxial detection. The tilt of Ma and Mb can be calibrated using active optics technology before each observation.

Coaxial detection requires a theodolite and specified marks in the optical path, which include the Shack-

Hartmann centre (SHC) of the Ma mirror, two cross marks at the top (TS) and bottom (BS) of the slope between Ma and Mb, the centre of the Mb (CMB) mirror, and cross marks on the top (TF) and bottom (TB) of the focal plane. The centre of Mb is marked by pasting the reflector.

Table 9 Coaxial detection of LAMOST. The standard azimuth angle value of SHC is 180° . The standard azimuth angle of the other marks is 0° . The standard azimuth angle between SHC and CMB is -25° .

Detection position	Azimuth angle	Vertical angle
SHC	$180^{\circ}0'13.3''$	$-25^{\circ}0'0.2''$
TS	$-0^{\circ}0'2.3''$	$-12^{\circ}25'18.9''$
BS	$-0^{\circ}0'1.7''$	$-4^{\circ}3'25.1''$
CMB	$0^{\circ}0'11''$	$25^{\circ}0'3''$
TF	$-0^{\circ}0'14.8''$	$-25^{\circ}0'12.1''$
BF	$-0^{\circ}0'57.2''$	$-25^{\circ}0'35.4''$

The theodolite is set on the line between the SHC of the Ma mirror and the cross mark of the bottom slope. Then, the azimuth angle and vertical angle of the marks is detected.

The theodolite is set on the line between the SHC of the Ma and the cross mark of the bottom slope. The result is shown in Table 9.

The results of the coaxial detection show that the Ma, Mb, and focal panel coaxial detection accuracy are within 13 arcsec, and have less effect on focus surface attitude detection.

At the start of observation or before changing the survey area: when the target central star is near the culmination, we adjust the telescope focal plane position from positive direction defocus (toward Ma) to negative direction defocus (toward Mb). Imaging in the 20, 21, 22, 23, 24, 25, -20, -21, -22, -23, -24, -25 mm positions respectively, eight acquisition cameras are used to capture the image. The duration of the experiment was limited to within 20 minutes to reduce the effect of star position on the size of the image.

Although LAMOST is equipped with eight acquisition cameras that produce eight sets of data, it was a challenge to find multiple nonoverlapping bright stars for detection testing and LAMOST tracking requirement only guarantee four cameras have a bright star. In the March 2019 test, we were able to obtain three suitable targets. The results of image size and image detection are shown in Table 10 and Figure 8.

No.0 detector corresponds to the -1.5/1.5 FOV simulation, while the fourth detector corresponds to the 0/2.5 FOV simulation and the sixth detector corresponds to the 0/-2.5 FOV simulation.

A right-handed coordinate system is constructed, in which the direction from Mirror Mb to Ma is set as the positive direction of the Z-axis, horizon direction is set as X-axis, vertical direction is set as Y-axis. Comparing the experiment data with the look-up table, the test results show that No.0, No.4 and No.6 detectors have deviations of $-179 \ \mu m \sim -87.9 \ \mu m$, 12.7 μm , and 102.3 μm ,

Defocus distance	0 CCD	4 CCD	6 CCD
-25	235.098	195.978	202.500
-24	224.963	187.634	193.336
-23	215.082	182.261	184.632
-22	204.364	173.729	176.664
-21	192.835	166.906	166.464
-20	185.727	158.908	158.835
20	179.343	164.032	168.807
21	188.298	172.645	176.893
22	198.329	180.548	186.920
23	209.109	189.304	194.478
24	219.508	196.662	203.660

Table 10 The detection results of three CCD cameras and defocus distance has show good linearity. Using the look-up table, we give the three-position focal point deviation.

respectively. According to the geometric relationship, there is a reverse $56.2 \sim 97.9$ arcsec tilt around the horizontal axis and a positive 10.2 arcsec tilt around the vertical axis.

230.452

440.4

205.187

855.0

211.806

-716.8

5 CONCLUSIONS

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focal position

In this study, we used eight acquisition cameras to detect the focal plane attitude of LAMOST. While applying the focal plane defocus imaging method to detect focal plane attitude, we considered the influence of Ma and focal plane rotation and the collimation of the optical path during the tracking process. By limiting the experimental observation time and detecting near the culmination, we effectively controlled the influence of Ma and focal plane rotation.

The estimated focus position obtained by our method deviated from the focal surface. To correct the deviation, we used a look-up table method. The focus position detection accuracy achieved by the simulation experiment reached 30 arcsec.

Before detection of the focal plane attitude of LAMOST in our experiment, we calibrated the collimation of the optical path using a theodolite. In addition, we calibrated the tilt of the Ma and Mb mirrors by active optics to reduce the influence of tilt of Ma and Mb on the attitude detection of the focal plane.

According to the experimental results and the geometric relationship, there is a reverse $56.2 \sim 97.9$ arcsec tilt around the horizontal axis and a positive 10.2 arcsec tilt around the vertical axis.

Additional experimental observations are required to verify the reliability of our method. Our method aims to decrease the effect of attitude detection on normal observations. At the same time, this method can be applied to focal plane attitude detection in other large-field telescopes. 181 - 10

Acknowledgements We would like to express our gratitude to the LAMOST work group and for the help of VISTA's Gavin Dalton.

This work is supported by the National Natural Science Foundation of China (Grant No. U1931207). The Guo Shou Jing Telescope (or LAMOST) is a National Major Scientific Project built by the Chinese Academy of Sciences. Funding for the project has been provided by the National Development and Reform Commission. LAMOST is operated and managed by National Astronomical Observatories, Chinese Academy of Sciences.

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